# Ultra-High Power and Efficiency Space Traveling-Wave Tube Amplifier Power Combiner With Reduced Size and Mass for NASA Missions

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Abstract—In the 2008 IEEE Microwave Theory and Techniques Society International Microwave Symposium Digest version of our paper, recent advances in high power and efficiency space traveling-wave tube amplifiers for NASA's space-to-Earth communications are presented. The RF power and efficiency of a new K-band amplifier are 40 W and 50% and that of a new Ka-band amplifier are 200 W and 60%. An important figure-of-merit, which is defined as the ratio of the RF power output to the mass (W/kg) of a traveling-wave tube (TWT), has improved by a factor of 10 over the previous generation Ka-band devices. In this paper, a high power high efficiency Ka-band combiner for multiple TWTs, based on a novel hybrid magic-T waveguide circuit design, is presented. The measured combiner efficiency is as high as 90%. In addition, at the design frequency of 32.05 GHz, error-free uncoded binary phaseshift keying/quadrature phase-shift keying (QPSK) data transmission at 8 Mb/s, which is typical for deep-space communications, is demonstrated. Furthermore, QPSK data transmission at 622 Mb/s is demonstrated with a low bit error rate of  $2.4 \times 10^{-8}$ , which exceeds the deep-space state-of-the-art data rate transmission capability by more than two orders of magnitude. A potential application of the TWT combiner is in deep-space communication systems for planetary exploration requiring transmitter power on the order of a kilowatt or higher.

Index Terms—Amplifiers, magic-T, microwave power amplifiers, millimeter wave power amplifiers, millimeter wave tubes, power combiner, power conditioning, satellite communication, space technology, traveling-wave tubes (TWTs), waveguide.

#### I. INTRODUCTION

Communications signals transmitted from space-crafts orbiting the outer planets or from the vicinity of the lunar surface need to travel enormous distances to ground antennas. In addition, as the distance from the Sun increases, the amount of solar energy that is available for conversion to electrical power on board the spacecraft decreases. Hence, there is a need for developing reliable space amplifiers, which are capable of delivering ultra-high RF power, as well as operating with very high efficiency [1]. In addition, these amplifiers need to be lightweight and small in size. The two possible solutions to this problem are based on either solid-state or microwave vacuum electronics technology. Solid-state devices based

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on GaAs or GaN high electron mobility transistor (HEMT) technology, when characterized on-wafer using load-pull techniques, deliver only about 2–3 W of RF power at Ka-band frequencies [2]. Although these devices have efficiencies as high as 35%-40%, the overall efficiency of Ka-band high power solid-state power amplifiers (SSPAs) are on the order of 15%–20%, as described in the 2008 IEEE Microwave Theory and Techniques Society (IEEE MTT-S) International Microwave Symposium (IMS) Digest version of our paper [3]. The low efficiency results in thermal management issues, which impacts reliability. On the other hand, microwave vacuum electronics based devices for space applications such as helix traveling-wave tubes (TWTs) have demonstrated reliability at higher microwave frequencies with power output in the range of ten to several hundred watts and corresponding efficiency in the range of 40%–60% [4], [5].

Deep-space missions in the past have typically relied either on solar cells or radioactive thermal generators as a source of electrical power, which greatly limits the power available for communications. As an alternative, the National Aeronautics and Space Administration (NASA), Cleveland, OH, has been developing nuclear fission electric power and propulsion technologies, which would provide orders of magnitude more power for science instruments onboard the spacecraft. An example was the Prometheus project [6]. This would potentially make possible the support of high power communications systems with data transmission rates  $1000 \times$  higher (8–10 Mb/s) than that previously achieved (10 kb/s) from deep space. To enable such high data rates, several approaches to develop an onboard RF transmitter with an output power on the order of about 1 kW at Ka-band, including power combining, are being considered.

Communication from the vicinity of the Moon is considered to be near-Earth, while that from the planets of our solar system is considered to be from deep space. The corresponding downlink frequency band designated for near-Earth is 25.5-26.5 GHz (K-band) and for deep space communications is 31.8-32.3 GHz (Ka-band). In the 2008 IEEE MTT-S IMS Digest version of our paper [3], we presented the microwave performance of a new K-band helix TWT for near-Earth communications and also a new Ka-band helix TWT for deep-space communications. Both of these units were manufactured by L-3 Communications, Electron Technologies Inc. (ETI), Torrance, CA, under contracts from NASA Glenn Research Center (GRC).

In this paper, we present the development of a novel Ka-band power combiner architecture based on the use of

hybrid magic-T's for combining the power output from multiple TWTs in a binary configuration with the potential to achieve 1 kW [7]. This architecture avoids the need to develop a single ultra-high power TWT, which would be technically challenging. To the best of our knowledge, this work demonstrates the highest power combining achieved with a Ka-band magic-T. The TWTs utilized in the combiner experiment were also manufactured by L-3 Communications ETI, under a contract from NASA GRC. Our approach has the following unique features: first, the size and mass are small, which are desired features for space applications. Second, it has low insertion loss and excellent output amplitude/phase balance, which ensures high combining efficiency. Third, it has very high isolation between the two input ports, which allows a TWT to continue operating if the other fails. Fourth, the circuit is capable of operating over nearly 60% of the waveguide band, and hence, has wide bandwidth if needed for high data rates. Fifth, the ports are impedance matched, and hence, minimize reflection-introduced inter-symbol interference, which degrades bit error rate (BER). Sixth, the experiments reported in this paper have been carried out over the NASA's Deep Space Network (DSN) downlink Ka-band frequency range of 31.8–32.3 GHz.

#### II. TWT DESIGN AND MODELING

Until recently, the design of TWTs was mainly done through trial and error, which was time consuming and expensive. The advances made in desktop computing and electromagnetic simulation/optimization tools have enabled the first pass design success of modern TWTs. These include the U.S. Naval Research Laboratory's CHRISTINE 3-D Code for high efficiency slow-wave interaction circuit design and MICHELLE 3-D Code for multistage depressed collector design [8]-[10]. In addition, efficient thermal modeling/simulation tools are also available, which have enhanced the power handling capability of TWTs by integrating efficient conduction cooled packages. Furthermore, advances in materials technology have resulted in lightweight temperature-stable high B-H (magnetic flux-magnetic field) product samarium cobalt permanent magnets, which are used in focusing the electron beam. Moreover, advances in tungsten/osmium cathode technology have resulted in cathode lifetimes exceeding 20 years in space [11].

# III. K- AND Ka-BAND SPACE TWT REQUIREMENTS AND MEASURED PERFORMANCE

#### A. L-3 ETI Model 9835H K-Band TWT

The *K*-band TWT was developed for communications from the Lunar Reconnaissance Orbiter (LRO) spacecraft to Earth. This spacecraft is scheduled for launch in April 2009. The specifications for this TWT are presented in [3]. Three TWTs were manufactured and characterized and all three met full specification, resulting in first pass design success and 100% yield. The measured output power, overall efficiency, and saturated gain of the three TWTs are presented in [3].

#### B. L-3 ETI Model 999H Ka-Band TWT

This Ka-band TWT was developed for communications from a spacecraft orbiting Mars to NASA's DSN ground stations.

The power output and overall efficiency of the TWT are 100 W and 60%, respectively. This TWT also served as an engineering model to validate the advanced design codes described earlier in Section II and paved the way for the successful development of the space qualified 200-W TWT described below.

# C. L-3 ETI Model 999HA Ka-Band TWT

This Ka-band TWT was developed for communications from a spacecraft orbiting any of the outer planets/moons, such as Jupiter or its icy moons. The specifications for the model 999HA Ka-band TWT are presented in [3]. Three such TWTs were manufactured and characterized and all were able to demonstrate at least 180 W. The TWT S/N 202 was tested up to 250 W and for long service life space qualified at 200 W. The measured overall efficiency and saturated gain of this TWT exceeded 60% and 55 dB, respectively. The measured characteristics are presented in [3].

To transmit data from Jupiter at very high rates on the order of several megabits per second, the spacecraft transmitter power needs to be on the order of 1 kW. To achieve this kind of power level, the outputs from four of the above TWTs would be combined. Combining efficiency as high as 90% has been demonstrated at Ka-band frequencies for a two-way combiner.

Both the K- and Ka-band TWTs have four-stage depressed collector circuits for high efficiency. The collector circuits require high voltages, which are provided by a separate electronic power conditioner (EPC) attached by an umbilical power cord to the TWT.

#### IV. EPCs FOR TWTs

The model 2300HE 7-kV EPC is mated with the 40-W K-band TWT and the model 1693HC 14-kV EPC is mated with the 180-W Ka-band TWT to form two new TWT amplifiers (TWTAs). These EPCs are also manufactured by L-3 Communications ETI. These EPCs are highly reliable and efficient (90%) and can operate from either regulated or unregulated spacecraft bus voltages. A single EPC can be used to independently operate two TWTs. The masses and dimensions of the TWT and the EPC are presented in [3].

#### V. HIGH POWER COMBINER OPERATING PRINCIPLE

A two-way high power combiner based on a waveguide magic-T hybrid coupler is shown in Fig. 1(a). In this approach, the two TWTAs are coupled to Ports #1 and #4 (H- and E-plane ports along the plane of symmetry) of the magic-T, respectively. The incident power at Port #1 (from TWTA #1) is evenly split into two in-phase components between Ports #2 and #3. The power incident at Port #4 (from TWTA #2) is also evenly split, but with  $180^{\circ}$  phase difference between Ports #2 and #3. The magic-T is designed such that the electrical lengths between any port and the junction are identical. Thus, by the superposition principle, the power at Port #2 must add in phase, while an equal power at Port #3 must result in perfect cancellation. Consequently, Ports #2 and #3 are referred to as the sum and difference ports, respectively. This two-way combiner architecture can be extended in a binary configuration

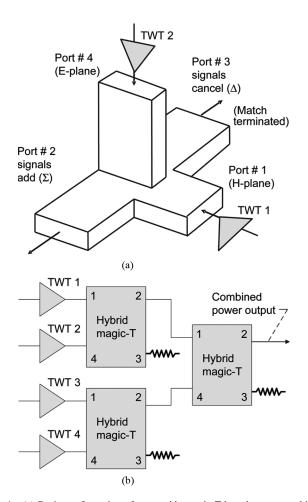


Fig. 1. (a) Basic configuration of waveguide magic-T based power combiner for two coherent TWTs. (b) Binary circuit architecture for combining  $2^n$  TWTs, where n is an integer (shown here for n=2).

to obtain 1 kW by combining  $2^n$  TWTAs, where n is an integer, as illustrated in Fig. 1(b).

## VI. HIGH POWER HYBRID MAGIC-T

The high power hybrid magic-T used in our demonstration was manufactured by Millitech Inc., Northampton, MA, and is shown in Fig. 2. The two input ports and the sum and difference ports are indicated. The magic-T is fabricated from aluminum and has a split block construction for ease of fabrication. In addition, steel guide pins are provided at all ports for precision alignment. The measured worst case insertion loss and voltage standing-wave ratio (VSWR) are about 0.45 dB and 1.47, respectively, and they occur at the high frequency end of the band. In addition, the measured amplitude and phase balance are within 0.27 dB and  $\pm 6^{\circ}$ , respectively. Using equations from [12], the calculated additional losses in combining efficiency due to the above amplitude and phase deviations are about 3.0% and 1.1%, respectively.

Lastly, measurements show that Ports #1 and #4 of the magic-T have very high isolation on the order of 30 dB or higher across the band of interest. Therefore, the magic-T was chosen over a hybrid 90° coupler, which has an isolation of

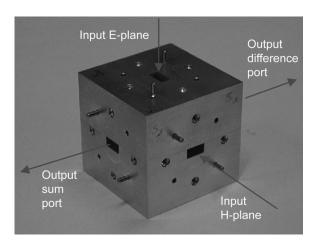


Fig. 2. Waveguide (WR-28) hybrid magic-T used in high power combiner circuit showing sum and difference output ports.

about 20 dB. The mass and size of the magic-T are about 100 g and  $50 \times 50 \times 50$  mm, respectively.

#### VII. POWER COMBINER MEASURED PERFORMANCE

The power combiner was successfully demonstrated with a pair of 100-W (L-3 ETI Model 999H, Serial Number 101 and 102) high efficiency Ka-band space TWTs. Fig. 3(a)–(c) shows the measured power output from the two TWTs and the magic-T combiner as a function of the input power to the TWT at the lower edge (31.8 GHz), center (32.05 GHz), and upper edge (32.3 GHz) of the NASA DSN 500-MHz-wide frequency band. It can be seen that the two TWTs are well balanced with the combined output power at about 3 dB (2×) greater. The AM-to-PM values for the two TWTs are less than 3.5 deg/dB at saturation. The phase changes at the output of the two TWTs when the input RF drive is increased from small signal (-20 dBm) to saturation (0 dBm) are 44° and 32°, respectively.

For these measurements, a microwave signal generator and a 3-dB power divider were used as the RF source to the two TWTs. In addition, one of the input arms to the TWTs included a variable attenuator and a variable phase shifter. The magic-T input powers were balanced with the variable attenuator. The magic-T input phases were balanced by adjusting the phase shifter for a minimum power reading at the difference port, which corresponds to a maximum combined power at the sum port. Consequently, the phase changes, AM-to-AM, and AM-to-PM characteristic differences of the TWTs did not impact the output power.

Fig. 4 shows the power combining efficiencies at the center (32.05 GHz) and  $\pm 250 \text{ MHz}$  band edge frequencies. The combining efficiencies were about 88.5% or greater. As shown in Fig. 4, the power combining efficiency is maintained over a broad range of input powers. The importance of this is that the power combiner is equally efficient at both the lower output power linear region, where multichannel commercial satellite transmitter TWTs usually operate, and the nonlinear saturation region, where deep-space communications systems typically operate for maximum TWT power and efficiency.

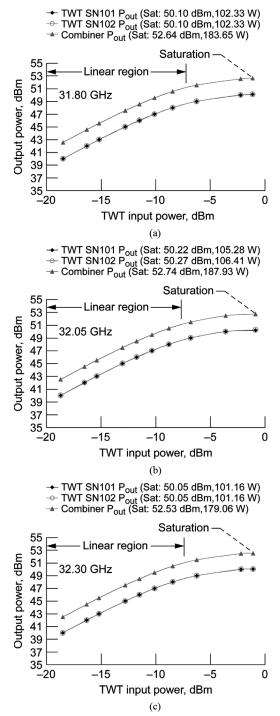


Fig. 3. Measured power output from individual TWTs (Model 999H) and magic-T combiner as function of input power to TWT. (a) 31.8 GHz. (b) 32.05 GHz. (c) 32.30 GHz.

In Fig. 4, at a fixed frequency, the difference in efficiency as a function of the total input power to the combiner is within  $\pm 0.5\%$ .

Improvements in performance are possible by optimizing the magic-T for efficiency and higher power handling capability over a narrow frequency range of specific interest. This was demonstrated with computer modeling of the magic-T and alternative hybrid junctions using Microwave Studio [13]. The folded E-plane magic-T appears to be the most promising of all the types that were investigated. A custom optimized magic-T

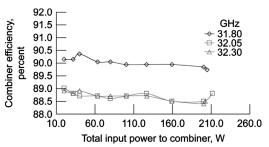


Fig. 4. Measured efficiency of magic-T based power combiner as function of total input power.

with about 500-MHz bandwidth and improved waveguide surface properties is estimated to have an insertion loss on the order of 0.25 dB [13].

#### VIII. COMBINER RELIABILITY

In the event of a single TWT failure, the combined power output would decrease by 6 dB. However, with appropriate placement of two waveguide switches, as shown in Fig. 5(a), the power loss would be only 3 dB. In this arrangement under normal operation, the waveguide switches allow the TWT outputs to be combined in the magic-T. If either one of the TWTs fail, then the other TWT is directly connected to the RF output. The sensor connected to Port #3 detects the power imbalance and commands the switches to reroute the TWT output. In addition, a ferrite circulator present between the RF output port and the antenna prevents the reflections from the antenna from influencing the imbalance sensor control.

With the addition of a backup TWT and appropriate placement of two waveguide switches, as shown in Fig. 5(b), the power loss would be zero. In this scheme under normal operation, TWT #1 and #2 are combined. In the event of a TWT failure, the standby TWT #3 can be switched in place of the failed TWT without loss in system performance. The circuits in Fig. 5(a) and (b) are work in progress, requiring additional efforts for performance optimization, and the design details are not available yet.

#### IX. BER CHARACTERIZATION

To demonstrate the TWT and combiner performance, we employed uncoded binary phase-shift keying (BPSK) and quadrature phase-shift keying (QPSK) bit streams. However, in an actual space-to-Earth link, both block coding, as well as convolution coding, would be used for error correction. In addition, QPSK allows the TWTA to operate in saturation, and consequently at the highest possible efficiency, which is a desired attribute in situations where prime power is scarce. BER tests were performed at the output of the magic-T combiner when the data rate was on the order of 8 and 622 Mb/s.

At 8 Mb/s, error-free (BER  $< 1 \times 10^{-9}$ ) uncoded BPSK and QPSK data transmission was observed. This was possible because the bandwidth occupied by the BPSK and QPSK signals were 8 and 4 MHz, respectively. Over these narrow bands, the rate of change in phase with frequency along the two TWT paths was small. The energy per bit/noise power spectral density ratio  $(E_b/N_0)$  was 21 dB at a BER of  $10^{-9}$  for a QPSK data rate of 8 Mb/s. Since there was no equalizer in the receiver, the BER

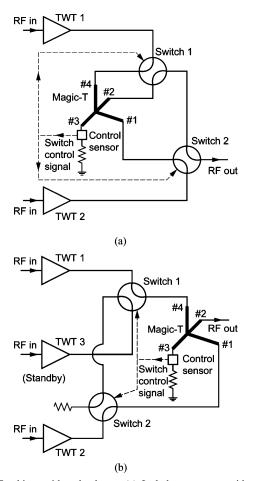


Fig. 5. Combiner with redundancy. (a) Includes two waveguide switches. (b) Includes two waveguide switches and standby TWT. Ports #1 through #4 correspond to ports of magic-T in Fig. 1(a).

was due primarily to inter-symbol interference. The 8 Mb/s is in the range of data rates proposed for some of the planetary exploration missions.

Future deep-space missions are calling for data rates approaching a gigabit per second at Ka-band. The bandwidth allocated at Ka-band is 500 MHz (31.8–32.3 GHz). A high data rate QPSK signal would occupy a significant portion of this bandwidth. However, in the characterization of our TWT combiner, we were limited by our test equipment, which could deliver only 622 Mb/s of QPSK data and required a bandwidth of 311 MHz. Initially, over this band, the rate of change in phase with frequency along the two TWT paths was large. This created a significant phase imbalance at the inputs to the magic-T combiner. The phase imbalance resulted in an observed reduction in signal amplitude of about 15 dB at the band edges. The above phase and amplitude changes made it impossible to detect the transmitted signal. By adding an appropriate dispersive component on one of the TWT paths, the disparity in electrical lengths and effect on RF amplitude was eliminated. The phase compensation was made by adding a section of a standard WR-28 waveguide at the input of one of the TWTs. With the waveguide in place, the measured rate of change of phase difference with frequency at the input ports to the magic-T was 0.147°/MHz compared to 0.991°/MHz before compensation and is presented in Fig. 6. The 0.147°/MHz rate

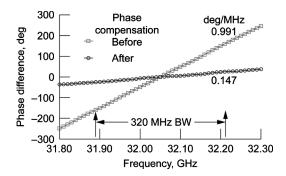


Fig. 6. Measured rate of change of phase difference at input ports to magic-T as function of frequency before and after phase compensation. For both cases, phase difference is set equal to  $0^{\circ}$  at center frequency of 32.05 GHz.

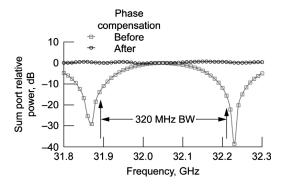


Fig. 7. Measured combined saturated relative power at sum port of magic-T as function of frequency before and after phase compensation. For both cases, relative power is set equal to 0 dB at 32.05 GHz.

of change of phase with frequency was observed out to at least 3 GHz (30.55–33.55 GHz).

Although further improvement is possible, this was adequate for a 622-Mb/s data transmission rate QPSK signal with low BER, which required only 311 MHz of the available bandwidth. Fig. 7 shows the measured combined saturated relative power at the sum port of the magic-T as a function of the frequency before and after compensation. As a result of this circuit modification, we were able to successfully demonstrate QPSK data transmission at 622 Mb/s with a low BER of  $2.4 \times 10^{-8}$ , which exceeds the data rate required for a typical deep space mission by more than two orders of magnitude.

### X. CONCLUSIONS AND DISCUSSIONS

The performance parameters of a K- and Ka-band TWTs for NASA's near-Earth and deep-space communications have been presented in the 2008 IEEE MTT-S IMS Digest version of our paper [3]. These results are state-of-the art and provide unprecedented performance. The model 999HA Ka-band 200-W TWT became available for the very first time in 2006. Prior to this development, the highest CW RF power produced by a U.S. manufactured space TWT at Ka-band and flown in space is the model 910H TWT. This TWT was manufactured by L-3 ETI for the Cassini-Huygens Mission with 10 W, 41% overall efficiency, and a mass of 0.75 kg [14]. The Ka-band model 999HA TWT increases the output power by a factor of 20 to 200 W, while only doubling the mass to 1.5 kg. Thus, a figure-of-merit

(FOM) defined as the ratio of the RF power output to the mass (W/kg) shows an increase of a factor of 10.

In this paper, a compact Ka-band high power high efficiency combiner based on a hybrid magic-T for combining the output from multiple TWTAs has been successfully demonstrated over NASA's DSN frequency band. At the design frequency of 32.05 GHz, error-free uncoded BPSK/QPSK data transmission at 8 Mb/s and QPSK data transmission at 622 Mb/s with a low BER of  $2.4 \times 10^{-8}$  was demonstrated. The 622-Mb/s data rate is significantly higher than the state-of-the-art, e.g., 5.247 Mb/s for the Mars Reconnaissance Orbiter (MRO) mission [15]. The measurements show that two 100-W TWTs can be combined with efficiency as high as 90%. Based on these measurements, the projected combiner efficiency of an eight-way combiner for eight 100-W model 999H TWTs to achieve about 800 W would be about 73%. On the other hand, if four 200-W model 999HA TWTs are used, then the projected combiner efficiency would be about 81%. The total mass of the combiner consisting of four 200-W TWTs [3], two EPCs [3], and a magic-T is about 15 kg and the total volume is equal to that of a cube with sides less than 28 cm.

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#### REFERENCES

- J. H. Booske, D. R. Whaley, W. L. Menninger, R. S. Hollister, and C. M. Armstrong, "Traveling-wave tubes," in *Modern Microwave and Millimeter-Wave Power Electronics*, R. J. Barker, Ed. New York: Wiley, 2005, ch. 4, pp. 171–245.
- [2] J. C. Freeman, E. G. Wintucky, and C. T. Chevalier, "High power high efficiency Ka-band power combiners for solid-state devices," NASA, Cleveland, OH, Tech. Memo NASA/TM 2006-214447, Nov. 2006. [Online]. Available: http://gltrs.grc.nasa.gov
- [3] R. N. Simons, J. D. Wilson, and D. A. Force, "High power and efficiency space traveling-wave tube amplifiers with reduced size and mass for NASA missions," in *IEEE MTT-S Int. Microw. Symp. Dig.*, Atlanta, GA, Jun. 15–20, 2008, pp. 319–322.
- [4] J. M. Weekley and B. J. Mangus, "TWTA versus SSPA: A comparison of on-orbit reliability data," *IEEE Trans. Electron Devices*, vol. 52, no. 5, pp. 650–652, May 2005.
- [5] E. F. Nicol, B. J. Mangus, and M. K. de Pano, "TWTA versus SSPA: A new look at Boeing fleet on-orbit reliability data and comparison factors," in *IEEE Int. Vacuum Electron. Conf.*, Monterey, CA, Apr. 25–27, 2006, pp. 61–62.
- [6] R. Taylor, "Prometheus Project final report," JPL, Pasadena, CA, Doc. 982–R120461, Oct. 1, 2005. [Online]. Available: http://www.ev-eryspec.com/NASA/NASA+-+JPL/index.php
- [7] E. G. Wintucky, R. N. Simons, K. R. Vaden, G. Lesny, and J. L. Glass, "High power combining of Ka-band TWTs for deep space communications," in *IEEE Int. Vacuum Electron. Conf.*, Monterey, CA, Apr. 25–27, 2006, pp. 63–64.
- [8] D. Chernin, T. M. Antonsen, Jr, B. Levush, and D. R. Whaley, "A three-dimensional multifrequency large signal model for helix traveling wave tubes," *IEEE Trans. Electron Devices*, vol. 48, no. 1, pp. 3–11, Jan. 2001
- [9] D. K. Abe, B. Levush, T. M. Antonsen, Jr, D. R. Whaley, and B. G. Danly, "Design of a linear C-band helix TWT for digital communications experiments using the CHRISTINE suite of large signal codes," *IEEE Trans. Plasma Sci.*, vol. 30, no. 3, pp. 1053–1062, Jun. 2002.

- [10] J. J. Petillo, E. M. Nelson, J. F. DeFord, N. J. Dionne, and B. Levush, "Recent developments to the MICHELLE 2-D/3-D electron gun and collector modeling code," *IEEE Trans. Electron Devices*, vol. 52, no. 5, pp. 742–748, May 2005.
- [11] J. D. Wilson, K. R. Vaden, E. G. Wintucky, D. A. Force, I. L. Krainsky, R. N. Simons, N. R. Robbins, W. L. Menninger, D. R. Dibb, and D. E. Lewis, "Advances in space traveling-wave tubes for NASA missions," Proc. IEEE (Special Issue on Technical Advances in Deep Space Communications and Tracking), vol. 95, no. 10, pp. 1958–1967, Oct. 2007.
- [12] S. E. Kubasek, D. M. Goebel, W. L. Menninger, and A. C. Schneider, "Power combining characteristics of backed-off traveling wave tubes for communications applications," *IEEE Trans. Electron Devices*, vol. 50, no. 6, pp. 1537–1542, Jun. 2003.
- [13] K. R. Vaden, "Computer aided design of Ka-band waveguide power combining architectures for interplanetary spacecraft," NASA, Cleveland, OH, Tech. Memo. NASA/TM 2006-214108, Feb. 2006. [Online]. Available: http://gltrs.grc.nasa.gov
- [14] A. N. Curren, J. A. Dayton, Jr., R. W. Palmer, K. J. Long, D. A. Force, C. E. Weeder, Z. A. Zachar, and W. L. Harvey, "The Cassini mission Ka-band TWT," in *IEEE Int. Electron Devices Meeting Tech. Dig.*, San Francisco, CA, Dec. 11–14, 1994, pp. 783–786.
- [15] C. D. Edwards, Jr. and R. DePaula, "Key telecommunications technologies for increased data return for future Mars exploration," *Acta Astronaut.*, vol. 61, no. 1–6, pp. 131–138, Jun.–Aug. 2007.



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finline components for millimeter-wave applications and X-band toroidal latching ferrite phase shifters for phased arrays. Since 1985, he has been with the Communications Technology Division (CTD), NASA Glenn Research Center (GRC), Cleveland, OH, where he is currently the Chief of the Electron and Opto-Electronic Device Branch. While with NASA GRC as an NRC Post-Doctoral Research Associate, he investigated optical control of MESFET and high electron-mobility transistor (HEMT) devices. As a Senior Engineer, he has been involved with high-temperature superconductivity, modeling of coplanar waveguide/stripline (CPW/CPS) discontinuities, CPW feed systems for printed antennas, linearly tapered slot arrays, packaging of monolithic microwave integrated circuits (MMICs), silicon carbide (SiC) diode mixers, RF microelectromechanical systems (MEMS), and GaN SSPAs. He is currently involved in the area of high-efficiency space TWT power combiners for communications from near-Earth, as well as deep space. He has authored or coauthored over 140 publications in refereed journals and international symposium proceedings. He authored Optical Control of Microwave Devices (Artech House, 1990). He authored the chapter "High Temperature Superconducting Coplanar Waveguide Microwave Circuits and Antennas" in Advances in High-Tc Superconductors (Trans-Tech Pub., 1993). He coauthored the chapter "Tapered Slot Antenna" in Advances in Microstrip and Printed Antennas (Wiley, 1997). He authored Coplanar Waveguide Circuits, Components, and Systems (Wiley, 2001). He also authored the chapter "Body-Sensor Network for Space and Military Applications" in Antennas and Propagation for Body-Centric Wireless Communications (MA: Artech House, 2006). He also coauthored the chapter "Transmission Lines and Waveguides" in Antenna Engineering Handbook (McGraw-Hill, 2007). He holds four U.S. patents.

Dr. Simons has organized workshops, chaired sessions, and served on the Technical Program Committees of several IEEE International Symposiums. He is a member of the Editorial Board of the IEEE TRANSACTIONS ON MICROWAVE THEORY AND TECHNIQUES and served as associate editor of the IEEE TRANSACTIONS ON ANTENNAS AND PROPAGATION. He was the recipient of the Distinguished Alumni Award. He has been a recipient of over 25 NASA Certificates of Recognitions/Tech Brief Awards and four Space Act Awards. He was a recipient of the NASA Public Service Medal (2001) and the NASA Group Achievement Honor Award (2006) for pioneering research and development of microwave printed antennas/distribution media and high power amplifiers, respectively. He was also a recipient of the R&D100 Award (2006) for a new TWTA, which was voted as one of the 100 most technically significant products for 2006.



**Edwin G. Wintucky** (M'95) received the B.Sc. degree in physics from The Ohio State University, Columbus, in 1958, and the M.S. degree in physics from Kent State University, Kent, OH, in 1980.

He has been with the National Aeronautics and Space Administration (NASA), Cleveland, OH, since 1964, where he has been involved in a variety of programs that includes electric propulsion and space communications, focusing on cathode technology research and development, where he served as NASA's representative in that area. More recently,

he has been involved with the development of communications and remote sensing applications for high-power Ka-band TWTs that includes high power combining and dual-frequency Doppler radar.



**Jeffrey D. Wilson** (M'87–SM'97) received the B.S. degree in physics from Bowling Green State University, Bowling Green, OH, in 1976, and the M.S. and Ph.D. degrees in physics from the University of Illinois at Urbana-Champaign, in 1978 and 1983, respectively.

From 1984 tp 1985, he spent a year of postgraduate study with the Air Force Thermionic Electronics Research (AFTER) Program at the University of Utah. Since 1983, he has been a member of the Vacuum Electronics Research Group, National Aeronautics

and Space Administration (NASA) Glenn Research Center, Cleveland, OH. His research efforts have focused on computational techniques to enhance the design and performance of coupled-cavity, helical, and novel TWTs and to investigate the electromagnetic properties of left-handed metamaterials. He is currently involved with utilizing metamaterials to improve terahertz amplifier efficiency.



**Dale A. Force** (M'84) received the B.S. and M.S. degrees in physics from Michigan State University, East Lansing, in 1974 and 1976, respectively, and the M.A. degree in electronic engineering (under the AFTER Program) from the University of Utah, Salt Lake City, in 1986.

Since 1983, he has been with the National Aeronautics and Space Administration (NASA) Glenn Research Center (formerly NASA Lewis Research Center), Cleveland, OH. His early research concerned multistage depressed collector (MDC)

design. He currently manages the development of a *K*-band TWTA for the Lunar Reconnaissance Orbiter (LRO) and the Communications, Navigation, and Networking reConfigurable Testbed (CoNNeCT) Project.